Discrete Time Immersion and Invariance Adaptive Control for Systems in Strict Feedback Form

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Abstract— This paper presents a new design procedure for the adaptive stabilization (regulation) via state feedback for discrete-time nonlinear systems in parametric strict-feedback form. The algorithm utilizes discrete-time adaptive backstepping in the controller construction together with a new method for the parameter estimator design. This approach provides a recursive construction which guarantees boundness of the closed-loop trajectories and global convergence to the origin of the state of the closed-loop system. The performance of the proposed method are illustrated by simulations.

I. INTRODUCTION

Adaptive control techniques have been extensively reported in the literature for continuous-time nonlinear systems, see for example [1], [2], [3]. Likewise for discrete-time systems there are several results. Lyapunov-based designs are reported in [4], [5]. A preliminary study proposing a "lookahead" adaptive backstepping design for a class of discretetime strict-feedback systems without using Lyapunov functions has been presented in [6], see also [7]. In [8], [9] a recursive design scheme different from standard backstepping has been proposed. Finally, [10] considers direct adaptive control for systems with matched uncertainties and [11] describes a periodic adaptive control approach. Most of the recent works deal with the robust adaptive control problems. The problem is studied for systems in strict-feedback form perturbed by a class of nonlinear uncertainties in [12], where local stability is proved without using Lyapunov functions. This method is improved in [13]. A robust backstepping adaptive controller design for nonlinear discrete-time systems in parametric-strict-feedback form without overparametrization is given [14], [15]. In [16], [17] robust asymptotic and output tracking adaptive control problems are considered for strict feedback SISO systems. Finally, in [18] some results on the robust control of first-order nonlinear systems with both parametric and non-parametric uncertainties are presented, and in [19], [20], [21] the solution to some discrete-time adaptive control problems for a class of strict feedback systems with unknown control directions are developed.

In this paper an adaptive controller design via state-feedback for the adaptive regulation of linearly parametrised discretetime systems in feedback form is presented. The novelty of the method is that it allows for stable dynamics to be assigned to the parameter estimation error, extending the idea introduced in [3], [22] for continuous-time systems to the discrete-time setting.

By analogy to the continuous-time method, the discrete-time parameter estimation error dynamics include a free function which allows to "shape" the parameter estimation error dynamics, which can be regarded as linear time-varying.

The rest of the paper is organized as follows. In Section II preliminary results on the estimator and adaptive control design for a class of linearly parametrized nonlinear systems are given. The main results are presented in Section III. Finally, the method is applied to the problem of wing rock elimination in high-performance aircrafts and simulation results are presented in Section IV.

In the sequel, the notation $x_i^+(k) = x_i(k+1)$, $x_i^-(k) = x_i(k-1)$, $A^{+m}(x_1,...,x_n) = A(x_1^{+m},...,x_n^{+m})$ and $A^{-m}(x_1,...,x_n) = A(x_1^{-m},...,x_n^{-m})$, where $A(x_1,...,x_n) \in \Re^n \to \Re^p$ is used. Note that when convenient the index k is omitted.

II. PRELIMINARY RESULTS

Consider a class of discrete-time nonlinear systems described by the equation

$$x^{+} = f_0(x) + f_1(x)\theta + g(x)u,$$
(1)

where $x(k) \in \Re^n$ is the state vector, $u(k) \in \Re^m$ is the control vector, $\theta \in \Re^p$ is a vector of constant parameters and f_0 , f_1 and g are mappings of appropriate dimensions, with $f_0(0) = 0$ and $f_1(0) = 0$, and the problem of designing a discrete-time adaptive state feedback control law of the form

$$\hat{\theta}^+ = \alpha(x, \hat{\theta}), \quad u = \upsilon(x, \hat{\theta}),$$
 (2)

such that all trajectories of the closed-loop system are bounded and

$$\lim_{k \to \infty} x(k) = 0. \tag{3}$$

Additionaly, we may require that the equilibrium $(x, \hat{\theta}) = (0, \theta)$ be stable. To solve this problem define the parameter estimation error $z = \hat{\theta} - \theta + \check{\beta}(x^-, x)$ with $\check{\beta}(x^-, x) = \beta(x^-)x$. Note that $z^+ = \hat{\theta}^+ - \theta + \beta(x)x^+$, hence

$$z^{+}-z = \hat{\theta}^{+} - \hat{\theta} + \beta(x)x^{+} - \beta(x^{-})x$$

$$= \hat{\theta}^{+} - \hat{\theta} + \beta(x)f_{1}(x)\theta +$$

$$\beta(x)(f_{0}(x) + g(x)u) - \beta(x^{-})x \qquad (4)$$

$$= \hat{\theta}^{+} - \hat{\theta} + \beta(x)f_{1}(x)[\hat{\theta} + \beta(x^{-})x - z] +$$

$$\beta(x)(f_{0}(x) + g(x)u) - \beta(x^{-})x.$$

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Selecting the update law as

$$\hat{\theta}^{+} = \hat{\theta} - \beta(x)f_1(x)[\hat{\theta} + \beta(x^-)x] - \beta(x)(f_0(x) + g(x)u) + \beta(x^-)x,$$
(5)

yields the parameter estimation error dynamics

$$z^{+} = [I - \beta(x)f_{1}(x)]z.$$
(6)

Note that if the function $\beta(x)$ is such that

$$\bar{\sigma}[I - \beta(x)f_1(x)] < 1, \tag{7}$$

then the parameter estimation error converges to zero. This observation motivates the following preliminary result.

Proposition 1: Consider the nonlinear system (1). Suppose there exists a control law

$$u = v(x, \theta), \tag{8}$$

such that the zero equilibrium of the closed-loop system

$$x^{+} = f_0(x) + f_1(x)\theta + g(x)\upsilon(x,\theta), \qquad (9)$$

is globally asymptotically stable. Let $\beta(x)$ be such that

$$\bar{\sigma}[I - \beta(x)f_1(x)] \le 1, \tag{10}$$

and such that the trajectories of the system

$$z^{+} = [I - \beta(x)f_{1}(x)]z, \qquad (11)$$

satisfy, for all x,

$$\lim_{k \to \infty} g(x) [\upsilon(x, \theta + z) - \upsilon(x, \theta)] = 0.$$
(12)

Then all trajectories of the closed-loop system

$$x^{+} = f_{0}(x) + f_{1}(x)\theta + g(x)\upsilon(x,\theta_{est})$$

$$\hat{\theta}^{+} = \hat{\theta} - \beta(x)f_{1}(x)[\hat{\theta} + \beta(x^{-})x] -$$
(13)

$$\beta(x)(f_{0}(x) + g(x)u) + \beta(x^{-})x,$$

where $\theta_{est} = \hat{\theta} + \beta(x^{-})x = \theta + z$, are bounded and $\lim x(k) = 0$. \Box

III. MAIN RESULTS

In this section we consider discrete-time systems described by the equations

$$\begin{aligned}
x_1^+ &= x_2 + \Phi_1^T(x_1)\theta, \\
x_2^+ &= x_3 + \Phi_2^T(x_1, x_2)\theta, \\
&\vdots \\
x_i^+ &= x_{i+1} + \Phi_i^T(x_1, ..., x_i)\theta, \\
&\vdots \\
x_n^+ &= u + \Phi_n^T(x_1, ..., x_n)\theta,
\end{aligned}$$
(14)

where $x_i(k) \in \mathfrak{R}$, i = 1, ..., n, are the states, $u(k) \in \mathfrak{R}$ is the control input, $\Phi_i \in \mathfrak{R}^i \to \mathfrak{R}^p$, i = 1, ..., n, are mappings such that $\Phi_i(0) = 0$ and $\theta \in \mathfrak{R}^p$ is a vector of unknown constant parameters.

The objective is to design a discrete-time adaptive controller described by equations of the form (2) such that all closedloop trajectories are bounded and

$$\lim_{k \to \infty} (x_1(k) - x_1^*(k)) = 0, \tag{15}$$

where $x_1^*(k)$ is a given reference signal. This objective is achieved in two steps, as detailed hereafter.

A. Estimator Design

To begin with we design a stable estimator for the parameter θ . Let, similarly to Section II,

$$z_i = \hat{\theta}_i - \theta + \check{\beta}_i, \quad i = 1, .., n,$$
(16)

where the functions $\check{\beta}_i$ are defined as

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$$\vec{\beta}_i(x_1^-, ..., x_i^-, x_i) = \beta_i(x_1^-, ..., x_i^-)x_i,$$
(17)

hence

and

$$z_i = \hat{\theta}_i - \theta + \beta_i(x_1^-, \dots, x_i^-)x_i, \qquad (18)$$

$$z_{i}^{+} = \hat{\theta}_{i}^{+} - \theta + \beta_{i}(x_{1}, ..., x_{i})x_{i}^{+}.$$
 (19)

As a result

$$z_{i}^{+} - z_{i} = \hat{\theta}_{i}^{+} - \hat{\theta}_{i} + \beta_{i}(x_{1}, ..., x_{i})x_{i}^{+} - \beta_{i}(x_{1}^{-}, ..., x_{i}^{-})x_{i}$$

$$= \hat{\theta}_{i}^{+} - \hat{\theta}_{i} + \beta_{i}(x_{1}, ..., x_{i})[x_{i+1} + \Phi_{i}^{T}(x_{1}, ..., x_{i})\theta]$$

$$-\beta_{i}(x_{1}^{-}, ..., x_{i}^{-})x_{i} \qquad (20)$$

$$= \hat{\theta}_{i}^{+} - \hat{\theta}_{i} - \beta_{i}(x_{1}^{-}, ..., x_{i}^{-})x_{i} + \beta_{i}(x_{1}, ..., x_{i}) \times$$

$$[x_{i+1} + \Phi_{i}^{T}(x_{1}, ..., x_{i})[\hat{\theta}_{i} + \beta_{i}(x_{1}^{-}, ..., x_{i}^{-})x_{i} - z_{i}]],$$

and selecting

$$\hat{\theta}_{i}^{+} = \hat{\theta}_{i} + \beta_{i}(x_{1}^{-}, ..., x_{i}^{-})x_{i} - \beta_{i}(x_{1}, ..., x_{i}) \times [x_{i+1} + \Phi_{i}^{T}(x_{1}, ..., x_{i})[\hat{\theta}_{i} + \beta_{i}(x_{1}^{-}, ..., x_{i}^{-})x_{i}]], (21)$$

yields

$$z_i^+ = [I - \beta_i(x_1, ..., x_i) \Phi_i^T(x_1, ..., x_i)] z_i.$$
(22)

Let

$$\boldsymbol{\beta}_i = [\boldsymbol{I} + \boldsymbol{\Phi}_i \boldsymbol{\Phi}_i^T]^{-1} \boldsymbol{\Phi}_i, \qquad (23)$$

and note that $[I - \beta_i(x)]$

$$I - \beta_i(x_1, ..., x_i) \Phi_i^T(x_1, ..., x_i)] = [I + \Phi_i \Phi_i^T]^{-1}, \qquad (24)$$

hence

$$z_i^+ = [I + \Phi_i \Phi_i^T]^{-1} z_i, \quad i = 1, ..., n.$$
(25)

Consequently, the following statement holds.

Proposition 2: Consider the system (25). The zero equilibrium of the system is (uniformly) stable and $\Phi_i^T(x_1(k),...,x_i(k))z_i(k)$ is a l^2 signal. In addition, if $x_j(k)$, j = 1,...,i, is bounded, then

$$\lim_{k \to \infty} \Phi_i^T(x_1(k), ..., x_i(k)) z_i(k) = 0.$$
 (26)

B. Controller Design

In this section an adaptive controller solving the considered control problem is designed.

Theorem 1: Consider the system (14), and the control law

$$u = x_1^{*+n} - \sum_{m=1}^n (\Phi_{mest}^{+(n-m)})^T \theta_{mest}^{+(n-m)}, \qquad (27)$$

where x_1^* is the reference signal to be tracked, $\theta_{iest} = \hat{\theta}_i + \beta_i(x_1^-, ..., x_i^-)x_i$, with $\hat{\theta}_i$ obtained using the update law (21) and β_i given by (23); $\Phi_{mest}^{+(i-m-1)} = \Phi_m(x_{1est}^{+(i-m-1)}, ..., x_{iest}^{+(i-m-1)})$, i = 2, ..., n, with $x_{mest} = x_m$ and $x_{mest}^{+(i-m-1)}$, (i-m-1) > 0, estimated (i-m-1)-step future values of x_m obtained by recursively evaluating and utilizing θ_{mest}^{+k} , k = 1, ..., (i-m-1) in equations (14). Then the closed-loop system is such that all trajectories of the closed-loop system are bounded and

$$\lim_{k \to \infty} (x_1(k) - x_1^*(k)) = 0.$$
(28)

Moreover the equilibrium

$$\tilde{x}_i = 0, \quad z_i = 0, \quad i = 1, ..., n,$$
 (29)

where

$$\begin{aligned} \tilde{x}_i &= x_i - x_i^*, \quad i = 1, ..., n, \\ x_i^* &= x_1^{*+(i-1)} - \sum_{m=1}^{i-1} (\Phi_{mest}^{+(i-m-1)})^T \theta_{mest}^{+(i-m-1)}, \quad i = 2, ..., n, \end{aligned}$$

is Lyapunov stable. \Box

Remark : The control law *u* given in (27) consists of "look-ahead" values x_{iest}^+ of the state variable x_i . Note that

$$x_i^+ = x_{i+1} + \Phi_i^T(x_1, ..., x_i)\theta,$$
 (30)

$$x_{iest}^+ = x_{i+1} + \Phi_i^T(x_1, ..., x_i)\theta_{iest}.$$
 (31)

Hence

$$x_i^+ - x_{iest}^+ = -\Phi_i^T(x_1, ..., x_i)z_i.$$
 (32)

This, straightforwardly, implies

$$x_{iest}^{+m}|_{\Phi_{i}^{T}(x_{1},..,x_{i})z_{i}=0} = x_{i}^{+m}|_{\Phi_{i}^{T}(x_{1},..,x_{i})z_{i}=0},$$
(33)

since

$$\begin{aligned} x_{iest}^{+m}|_{\Phi_{i}^{T}(x_{1},...,x_{i})z_{i}=0} &= (x_{iest}^{+})^{+(m-1)}|_{\Phi_{i}^{T}(x_{1},...,x_{i})z_{i}=0}, \\ &= (x_{i}^{+})^{+(m-1)}|_{\Phi_{i}^{T}(x_{1},...,x_{i})z_{i}=0}, \\ &= x_{i}^{+m}|_{\Phi_{i}^{T}(x_{1},...,x_{i})z_{i}=0}. \end{aligned}$$
(34)

This analysis reveals that, as long as $\Phi_i^T(x_i)z_i = 0$, the value of the estimated variable x_{iest}^+ is equal to the true value of x_i^+ and this relation is also valid for all *i* and all future values. \Box

IV. EXAMPLE

The proposed methodology is applied to the problem considered in [22]: wing rock elimination in high-performance aircrafts. The discrete-time Euler model of the continuoustime system given in [22], which describes the motion of the wing, is given by

$$\begin{aligned} x_1^+ &= x_1 + T x_2, \\ x_2^+ &= x_2 + T x_3 + T \Phi^T(x_1, x_2) \theta, \\ x_3^+ &= x_3 + \frac{T}{\tau} (v - x_3), \end{aligned}$$
 (35)

where the states $x_1(k)$, $x_2(k)$ and $x_3(k)$ describe the roll angle, the roll rate and the aileron deflection angle, respectively, τ is the aileron time constant, v(k) is the control input, $\theta \in \mathbb{R}^5$ is a vector of unknown constant parameters,

$$\Phi(x_1, x_2) = [1, x_1, x_2, x_1 | x_2 |, x_2 | x_2 |],$$
(36)

and T is the sampling period. Applying the co-ordinates and input transformation

$$\hat{x}_{1} = x_{1},$$

$$\hat{x}_{2} = x_{1} + Tx_{2},$$

$$\hat{x}_{3} = x_{1} + 2Tx_{2} + T^{2}x_{3},$$

$$u = F + \frac{T^{3}}{\tau}v,$$
(37)

with

$$F = 2\hat{x}_3 - \hat{x}_2 + (1 - \frac{T}{\tau})(\hat{x}_3 + \hat{x}_1 - 2\hat{x}_2), \qquad (38)$$

yields the system in strict feedback form

$$\hat{x}_{1}^{+} = \hat{x}_{2},
\hat{x}_{2}^{+} = \hat{x}_{3} + \Phi_{2}^{T}(\hat{x}_{1}, \hat{x}_{2})\theta,
\hat{x}_{3}^{+} = u + \Phi_{3}^{T}(\hat{x}_{1}, \hat{x}_{2})\theta.$$
(39)

where

$$\Phi_2 = T^2 \Phi(\hat{x}_1, \hat{x}_2),
\Phi_3 = 2\Phi_2 = 2T^2 \Phi(\hat{x}_1, \hat{x}_2),$$
(40)

Simulations are carried out for T = 0.5 and the parameters $\tau = \frac{1}{15}$, $\theta_0 = [0, -26.72, 0.76485, -2.9225, 0]^T$ as in [1], and for $x_1^* = 0$. The initial conditions for the parameter estimator have been set to $\hat{\theta}(0) = 0$, and the initial state has been selected as $x(0) = [0.4, 0, 0]^T$. Figure 1 and Figure 2 show the time histories of the parameter estimation errors $z_2(k)$ and $z_3(k)$. Note that, consistently with the theory, $z_2(k)$ and $z_3(k)$ do not converge to zero. Figure 3 and Figure 4 show the time histories of the perturbations $\Phi_2(x)z_2(k)$ and $\Phi_3(x)z_3(k)$, respectively, which converge to zero. Finally, Figure 5 and Figure 6 display the time histories of the states $x_1(k), x_2(k)$, and $x_3(k)$ and of the control signal.

V. CONCLUSIONS AND FUTURE WORKS

In this paper a novel adaptive controller design method for the adaptive regulation of linearly parameterized discretetime systems in strict-feedback form has been developed. The control law is constructed using a backstepping-type procedure together with a novel parameter update law, which renders the zero equilibrium of the parameter estimation error system stable. This property is obtained, as in the continuoustime immersion and invariance approach, including a function of the state in the definition of the parameter estimation error. The results are illustrated on the Euler model of the system which describes the wing rock oscillation in highperformance aircraft. Further work to develop a stable parameter estimator which does not require overparametrization is under way.

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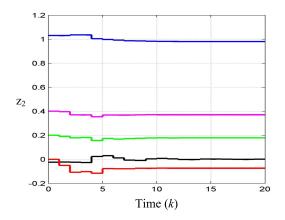


Fig. 1. The time history of the variable $z_2(k)$.

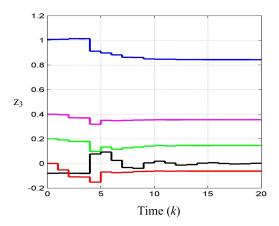


Fig. 2. The time history of the variable $z_3(k)$.

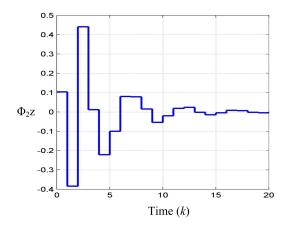


Fig. 3. The time history of the perturbation $\Phi_2(x(k))z_2(k)$.

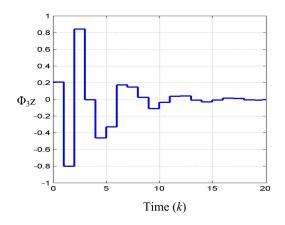


Fig. 4. The time history of the perturbation $\Phi_3(x(k))z_3(k)$.

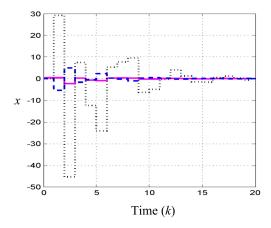


Fig. 5. The time histories of the states $x_1(k)$ (solid), $x_2(k)$ (dashed) and $x_3(k)$ (dotted).

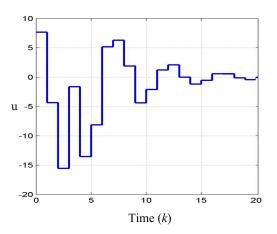


Fig. 6. The time history of the control signal.

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